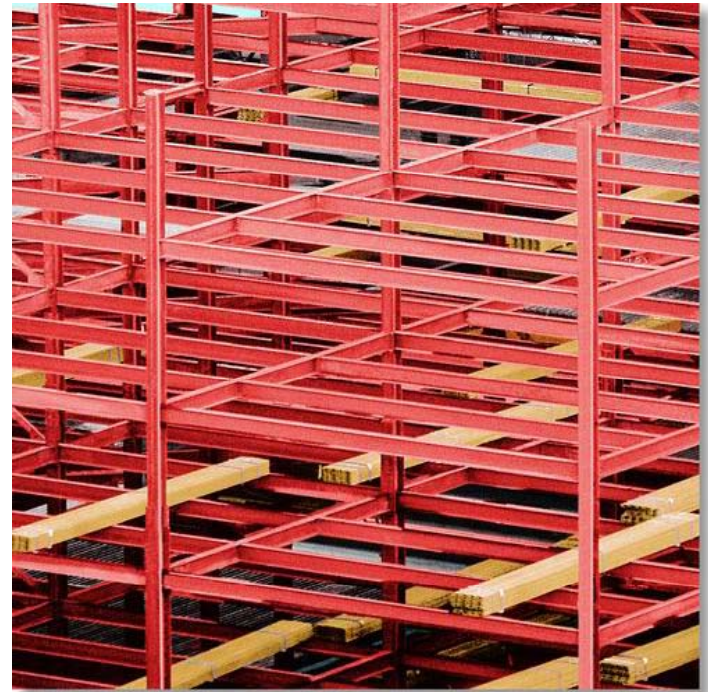


## 13. Buckling of Columns

### CHAPTER OBJECTIVES

- Discuss the behavior of columns.
- Discuss the buckling of columns.
- Determine the axial load needed to buckle an ideal column.
- Analyze the buckling with bending of a column.
- Discuss methods used to design concentric and eccentric columns.



## 13. Buckling of Columns

### CHAPTER OUTLINE

1. Critical Load
2. Ideal Column with Pin Supports
3. Columns Having Various Types of Supports
4. \*Design of Columns for Concentric Loading
5. \*Design of Columns for Eccentric Loading

## 13. Buckling of Columns

### 13.1 CRITICAL LOAD

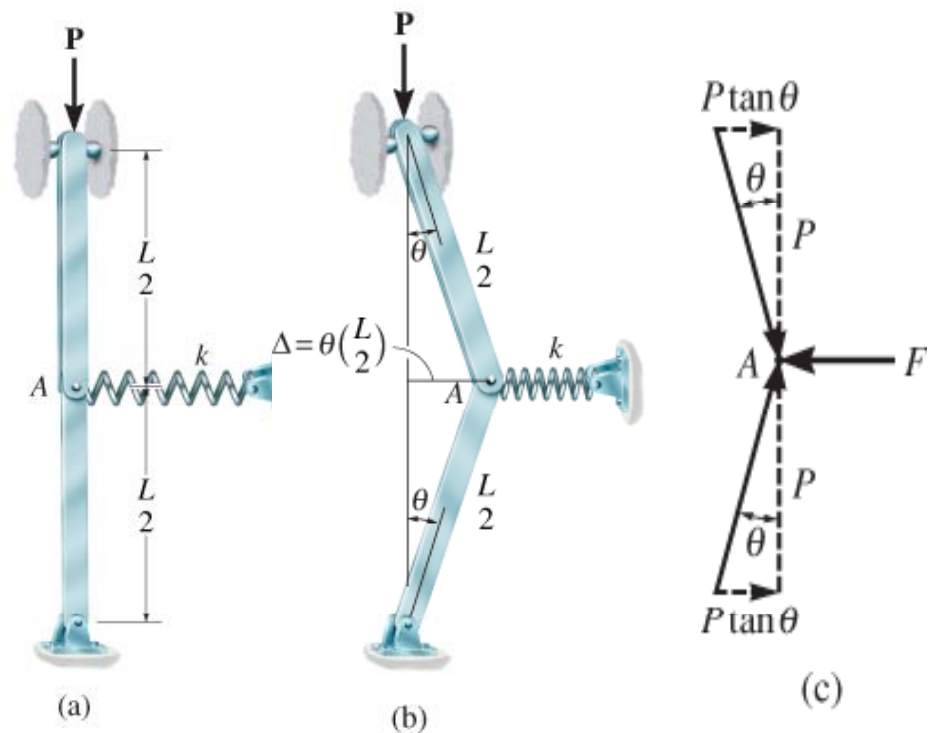
- Long slender members subjected to axial compressive force are called columns.
- The lateral deflection that occurs is called buckling.
- The maximum axial load a column can support when it is on the verge of buckling is called the critical load,  $P_{cr}$ .



## 13. Buckling of Columns

### 13.1 CRITICAL LOAD

- Spring develops restoring force  $F = k\Delta$ , while applied load  $\mathbf{P}$  develops two horizontal components,  $P_x = P \tan \theta$ , which tends to push the pin further out of equilibrium.
- Since  $\theta$  is small,  
 $\Delta = \theta(L/2)$  and  $\tan \theta \approx \theta$ .
- Thus, restoring spring force becomes  $F = k\theta L/2$ , and disturbing force is  $2P_x = 2P\theta$ .



## 13. Buckling of Columns

### 13.1 CRITICAL LOAD

- For  $k\theta L/2 > 2P\theta$ ,

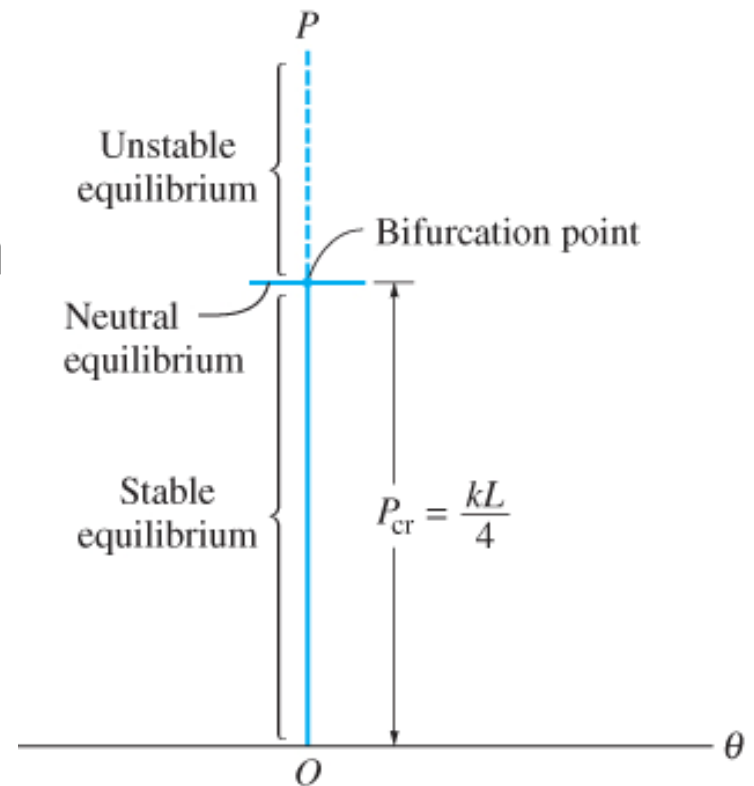
$$P < \frac{kL}{4} \quad \text{stable equilibrium}$$

- For  $k\theta L/2 < 2P\theta$ ,

$$P > \frac{kL}{4} \quad \text{unstable equilibrium}$$

- For  $k\theta L/2 = 2P\theta$ ,

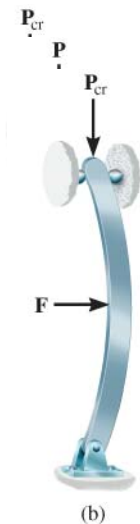
$$P_{cr} = \frac{kL}{4} \quad \text{neutral equilibrium}$$



## 13. Buckling of Columns

### 13.2 IDEAL COLUMN WITH PIN SUPPORTS

- An ideal column is perfectly straight before loading, made of homogeneous material, and upon which the load is applied through the centroid of the x-section.
- We also assume that the material behaves in a linear-elastic manner and the column buckles or bends in a single plane.



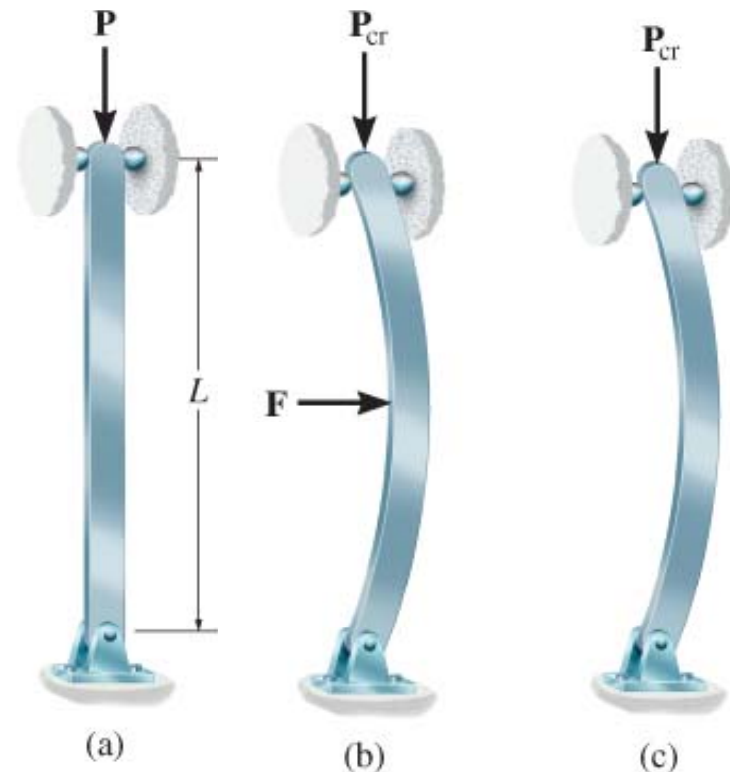
## 13. Buckling of Columns

### 13.2 IDEAL COLUMN WITH PIN SUPPORTS

- In order to determine the critical load and buckled shape of column, we apply Eqn 12-10,

$$EI \frac{d^2 v}{dx^2} = M \quad (13-1)$$

- Recall that this eqn assume the slope of the elastic curve is small and deflections occur only in bending. We assume that the material behaves in a linear-elastic manner and the column buckles or bends in a single plane.



## 13. Buckling of Columns

### 13.2 IDEAL COLUMN WITH PIN SUPPORTS

- Summing moments,  $M = -Pv$ , Eqn 13-1 becomes

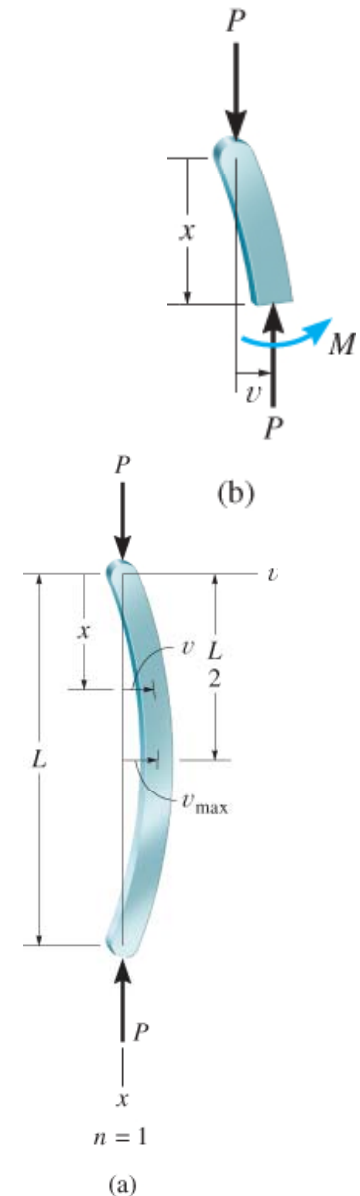
$$\frac{d^2v}{dx^2} + \left(\frac{P}{EI}\right)v = 0 \quad (13-2)$$

- General solution is

$$v = C_1 \sin\left(\sqrt{\frac{P}{EI}}x\right) + C_2 \cos\left(\sqrt{\frac{P}{EI}}x\right) \quad (13-3)$$

- Since  $v = 0$  at  $x = 0$ , then  $C_2 = 0$ .  
Since  $v = 0$  at  $x = L$ , then

$$C_1 \sin\left(\sqrt{\frac{P}{EI}}L\right) = 0$$





## 13. Buckling of Columns

### 13.2 IDEAL COLUMN WITH PIN SUPPORTS

- Disregarding trivial soln for  $C_1 = 0$ , we get

$$\sin\left(\sqrt{\frac{P}{EI}}L\right) = 0$$

- Which is satisfied if

$$\sqrt{\frac{P}{EI}}L = n\pi$$

- or

$$P = \frac{n^2 \pi^2 EI}{L^2} \quad n = 1, 2, 3, \dots$$

## 13. Buckling of Columns

### 13.2 IDEAL COLUMN WITH PIN SUPPORTS

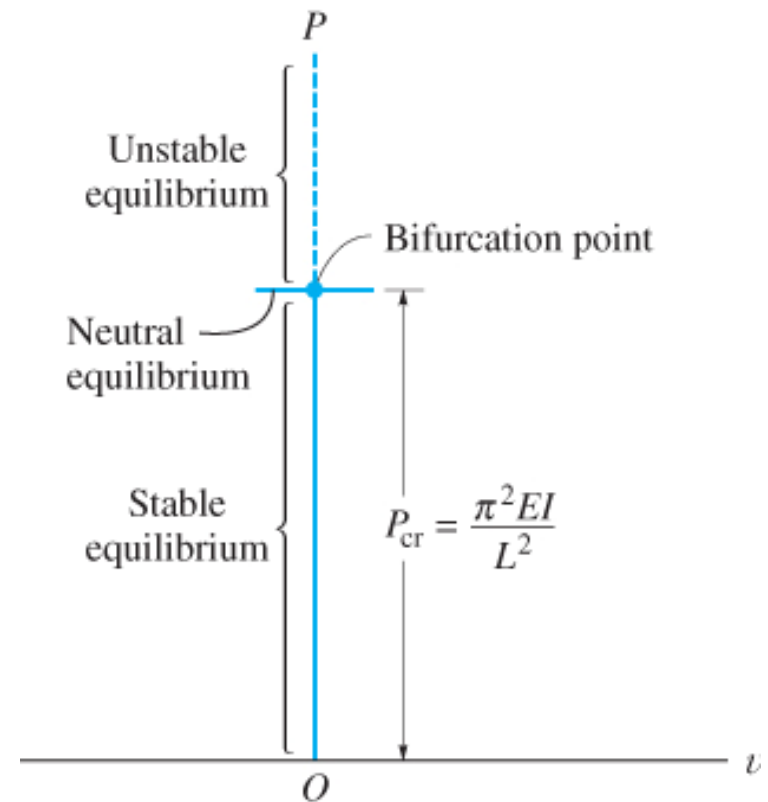
- Smallest value of  $P$  is obtained for  $n = 1$ , so critical load for column is

$$P_{cr} = \frac{\pi^2 EI}{L^2}$$

- This load is also referred to as the Euler load. The corresponding buckled shape is defined by

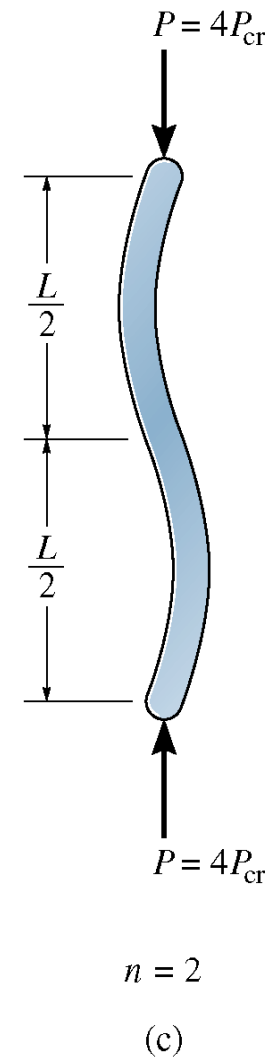
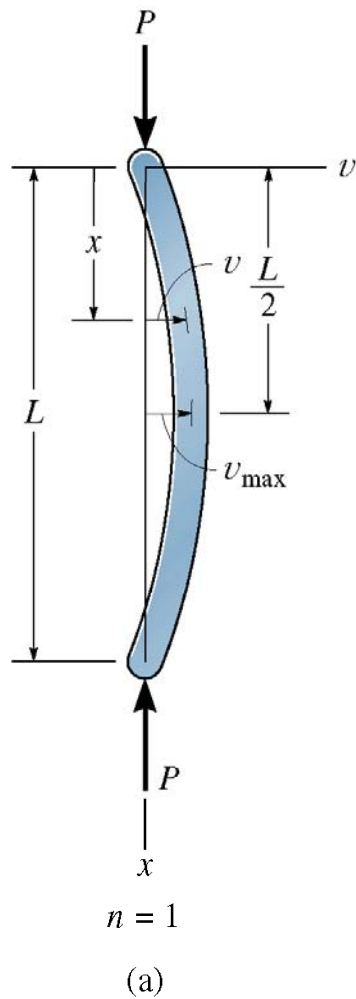
$$v = C_1 \sin \frac{\pi x}{L}$$

- $C_1$  represents maximum deflection,  $v_{\max}$ , which occurs at midpoint of the column.



## 13. Buckling of Columns

### 13.2 IDEAL COLUMN WITH PIN SUPPORTS



## 13. Buckling of Columns

### 13.2 IDEAL COLUMN WITH PIN SUPPORTS

- A column will buckle about the principal axis of the x-section having the least moment of inertia (weakest axis).
- For example, the meter stick shown will buckle about the  $a-a$  axis and not the  $b-b$  axis.
- Thus, circular tubes made excellent columns, and square tube or those shapes having  $I_x \approx I_y$  are selected for columns.



## 13. Buckling of Columns

### 13.2 IDEAL COLUMN WITH PIN SUPPORTS

- Buckling eqn for a pin-supported long slender column,

$$P_{cr} = \frac{\pi^2 EI}{L^2} \quad (13-5)$$

$P_{cr}$  = critical or maximum axial load on column just before it begins to buckle. This load must not cause the stress in column to exceed proportional limit.

$E$  = modulus of elasticity of material

$I$  = Least modulus of inertia for column's x-sectional area.

$L$  = unsupported length of pinned-end columns.

## 13. Buckling of Columns

### 13.2 IDEAL COLUMN WITH PIN SUPPORTS

- Expressing  $I = Ar^2$  where  $A$  is x-sectional area of column and  $r$  is the radius of gyration of x-sectional area.

$$\sigma_{cr} = \frac{\pi^2 E}{(L/r)^2} \quad (13-6)$$

$\sigma_{cr}$  = critical stress, an average stress in column just before the column buckles. This stress is an elastic stress and therefore  $\sigma_{cr} \leq \sigma_Y$

$E$  = modulus of elasticity of material

$L$  = unsupported length of pinned-end columns.

$r$  = smallest radius of gyration of column, determined from  $r = \sqrt{I/A}$ , where  $I$  is least moment of inertia of column's x-sectional area  $A$ .

## 13. Buckling of Columns

### 13.2 IDEAL COLUMN WITH PIN SUPPORTS

- The geometric ratio  $L/r$  in Eqn 13-6 is known as the slenderness ratio.
- It is a measure of the column's flexibility and will be used to classify columns as long, intermediate or short.

## 13. Buckling of Columns

### 13.2 IDEAL COLUMN WITH PIN SUPPORTS

#### IMPORTANT

- Columns are long slender members that are subjected to axial loads.
- Critical load is the maximum axial load that a column can support when it is on the verge of buckling.
- This loading represents a case of neutral equilibrium.



## 13. Buckling of Columns

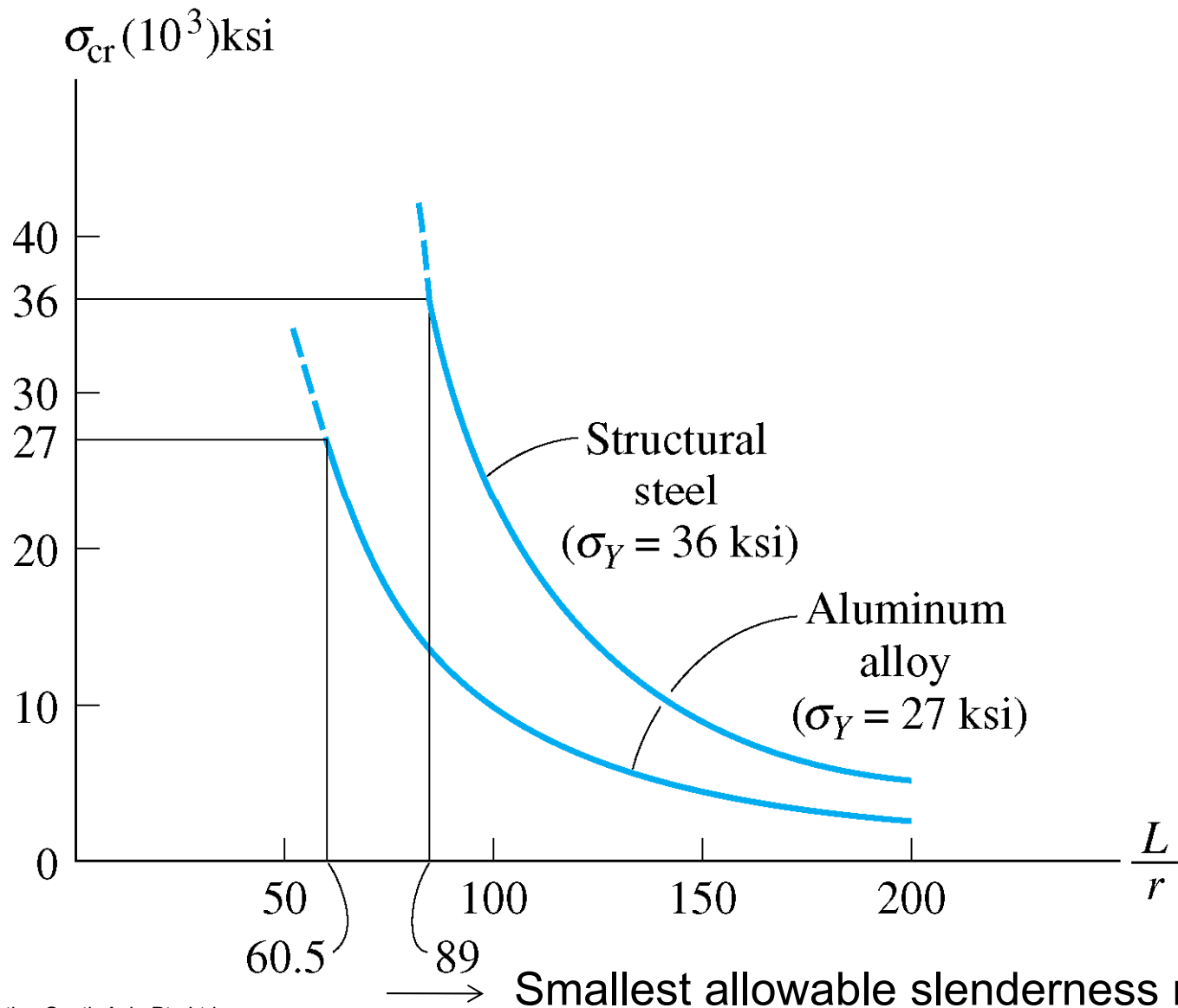
### 13.2 IDEAL COLUMN WITH PIN SUPPORTS

#### IMPORTANT

- An ideal column is initially perfectly straight, made of homogeneous material, and the load is applied through the centroid of the x-section.
- A pin-connected column will buckle about the principal axis of the x-section having the least moment of inertia.
- The slenderness ratio  $L/r$ , where  $r$  is the smallest radius of gyration of x-section. Buckling will occur about the axis where this ratio gives the greatest value.

## 13. Buckling of Columns

### 13.2 IDEAL COLUMN WITH PIN SUPPORTS

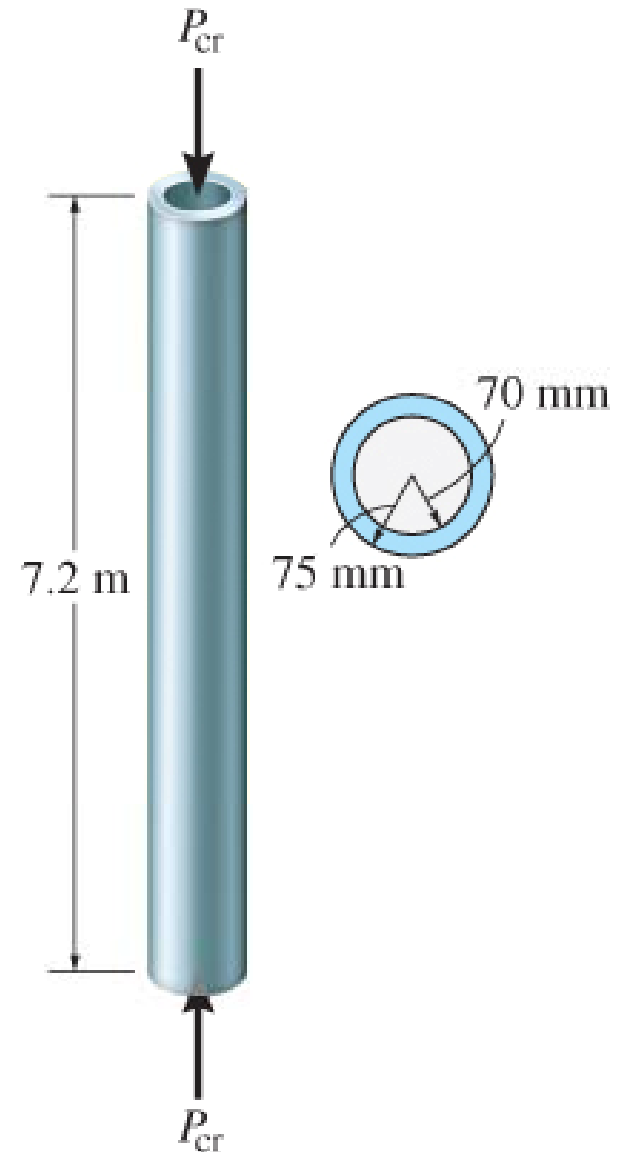


## 13. Buckling of Columns

### EXAMPLE (6<sup>th</sup> Ed.)

A 7.2-m long A-36 steel tube having the x-section shown is to be used a pin-ended column. Determine the maximum allowable axial load the column can support so that it does not buckle.

$$P_{cr} = \frac{\pi^2 EI}{L^2}$$



## 13. Buckling of Columns

### EXAMPLE (SOLN)

Use Eqn 13-5 to obtain critical load with  
 $E_{st} = 200 \text{ GPa}$ .

$$\begin{aligned} P_{cr} &= \frac{\pi^2 EI}{L_2} \\ &= \frac{\pi^2 \left[ 200(10^6) \text{ kN/m}^2 \right] - \frac{1}{4} \pi (70)^4 (1 \text{ m} / 1000 \text{ mm})^4}{(7.2 \text{ m})^2} \\ &= 228.2 \text{ kN} \end{aligned}$$

## 13. Buckling of Columns

### EXAMPLE (SOLN)

This force creates an average compressive stress in the column of

$$\begin{aligned}\sigma_{cr} &= \frac{P_{cr}}{A} = \frac{228.2 \text{ kN}(1000 \text{ N/kN})}{\left[\pi(75)^2 - \pi(70)^2\right] \text{ mm}^2} \\ &= 100.2 \text{ N/mm}^2 = 100 \text{ MPa}\end{aligned}$$

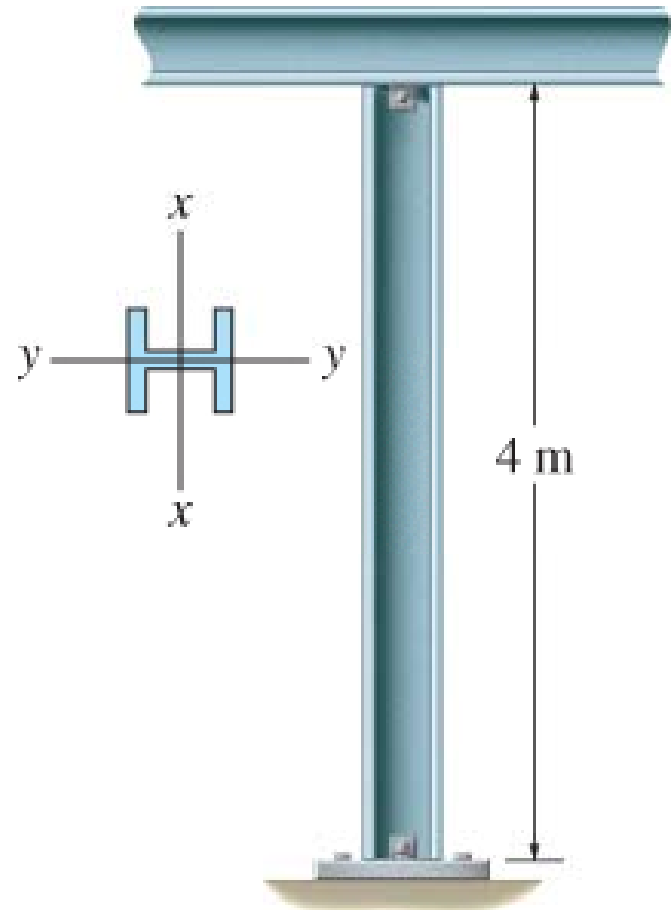
Since  $\sigma_{cr} < \sigma_Y = 250 \text{ MPa}$ , application of Euler's eqn is appropriate.

## 13. Buckling of Columns

### EXAMPLE 13.1

The A-36 steel W200×46 member shown is to be used as a pin-connected column. Determine the largest axial load it can support before it either begins to buckle or the steel yields.

$$P_{cr} = \pi^2 EI / L^2$$
$$\sigma = P_{cr} / A$$



## 13. Buckling of Columns

### EXAMPLE 13.1 (SOLN)

From table in Appendix B, column's x-sectional area and moments of inertia are  $A = 5890 \text{ mm}^2$ ,  $I_x = 45.5 \times 10^6 \text{ mm}^4$ , and  $I_y = 15.3 \times 10^6 \text{ mm}^4$ .

By inspection, buckling will occur about the  $y$ - $y$  axis.

Applying Eqn 13-5, we have

$$\begin{aligned} P_{cr} &= \frac{\pi^2 EI}{L_2} \\ &= \frac{\pi^2 [200(10^6) \text{ kN/m}^2] (15.3(10^4) \text{ mm}^4) (1 \text{ m} / 1000 \text{ mm})^4}{(4 \text{ m})^2} \\ &= 1887.6 \text{ kN} \end{aligned}$$

## 13. Buckling of Columns

### EXAMPLE 13.1 (SOLN)

When fully loaded, average compressive stress in column is

$$\begin{aligned}\sigma_{cr} &= \frac{P_{cr}}{A} = \frac{1887.6 \text{ kN}(1000 \text{ N/kN})}{5890 \text{ mm}^2} \\ &= 320.5 \text{ N/mm}^2\end{aligned}$$

Since this stress exceeds yield stress ( $250 \text{ N/mm}^2$ ), the load  $P$  is determined from simple compression:

$$\begin{aligned}250 \text{ N/mm}^2 &= \frac{P}{5890 \text{ mm}^2} \\ P &= 1472.5 \text{ kN}\end{aligned}$$



# 13. Buckling of Columns

## COLUMNS HAVING VARIOUS TYPES OF SUPPORTS

$$EI \frac{d^2 v}{dx^2} = P(\delta - v)$$

$$\frac{d^2 v}{dx^2} + \frac{P}{EI} v = \frac{P}{EI} \delta \quad (13-7)$$

Unlike Eq. 13-2, this equation is nonhomogeneous because of the nonzero term on the right side. The solution consists of both a complementary and a particular solution, namely,

$$v = C_1 \sin\left(\sqrt{\frac{P}{EI}} x\right) + C_2 \cos\left(\sqrt{\frac{P}{EI}} x\right) + \delta$$

The constants are determined from the boundary conditions. At  $x = 0$ ,  $v = 0$ , so that  $C_2 = -\delta$ . Also,

$$\frac{dv}{dx} = C_1 \sqrt{\frac{P}{EI}} \cos\left(\sqrt{\frac{P}{EI}} x\right) - C_2 \sqrt{\frac{P}{EI}} \sin\left(\sqrt{\frac{P}{EI}} x\right)$$

At  $x = 0$ ,  $dv/dx = 0$ , so that  $C_1 = 0$ . The deflection curve is therefore

$$v = \delta \left[ 1 - \cos\left(\sqrt{\frac{P}{EI}} x\right) \right] \quad (13-8)$$

Since the deflection at the top of the column is  $\delta$ , that is, at  $x = L$ ,  $v = \delta$ , we require

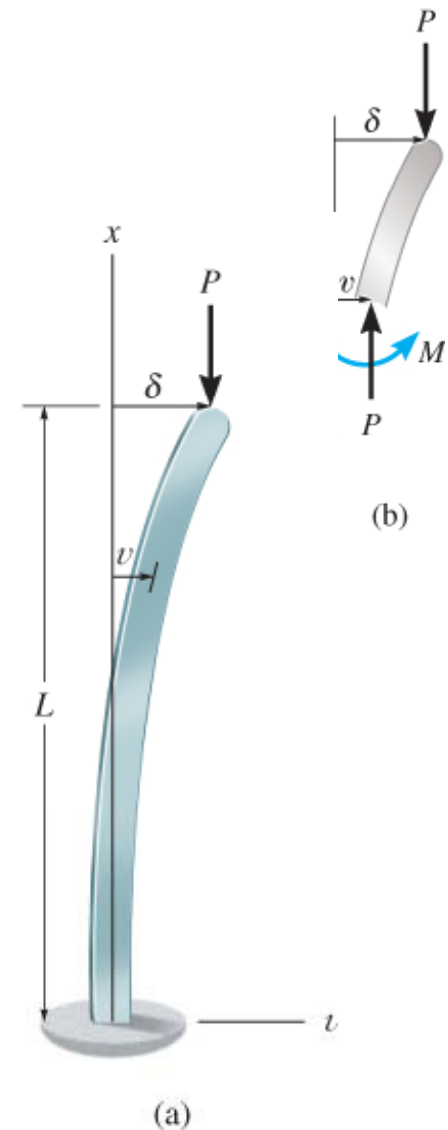
$$\delta \cos\left(\sqrt{\frac{P}{EI}} L\right) = 0$$

The trivial solution  $\delta = 0$  indicates that no buckling occurs, regardless of the load  $P$ . Instead,

$$\cos\left(\sqrt{\frac{P}{EI}} L\right) = 0 \quad \text{or} \quad \sqrt{\frac{P}{EI}} L = \frac{n\pi}{2}, n = 1, 3, 5, \dots$$

The smallest critical load occurs when  $n = 1$ , so that

$$P_{cr} = \frac{\pi^2 EI}{4L^2} \quad (13-9)$$



## 13. Buckling of Columns

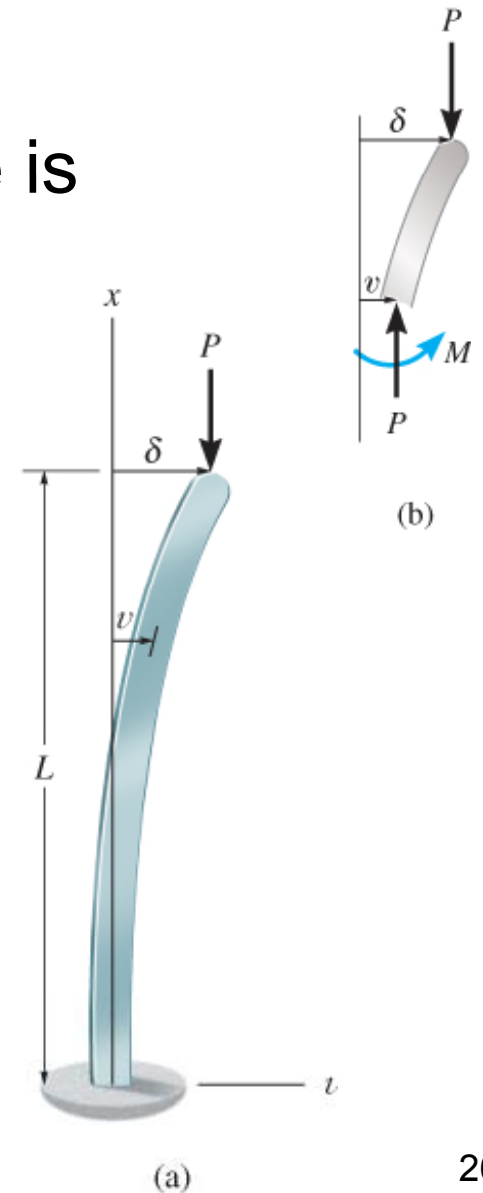
### 13.3 COLUMNS HAVING VARIOUS TYPES OF SUPPORTS

- From free-body diagram,  $M = P(\delta - v)$ .
- Differential eqn for the deflection curve is

$$\frac{d^2v}{dx^2} + \frac{P}{EI}v = \frac{P}{EI}\delta \quad (13-7)$$

- Solving by using boundary conditions and integration, we get

$$v = \delta \left[ 1 - \cos \left( \sqrt{\frac{P}{EI}} x \right) \right] \quad (13-8)$$



## 13. Buckling of Columns

### 13.3 COLUMNS HAVING VARIOUS TYPES OF SUPPORTS

- Thus, smallest critical load occurs when  $n = 1$ , so that

$$P_{cr} = \frac{\pi^2 EI}{4L^2} \quad (13-9)$$

- By comparing with Eqn 13-5, a column fixed-supported at its base will carry only one-fourth the critical load applied to a pin-supported column.

## 13. Buckling of Columns

### 13.3 COLUMNS HAVING VARIOUS TYPES OF SUPPORTS

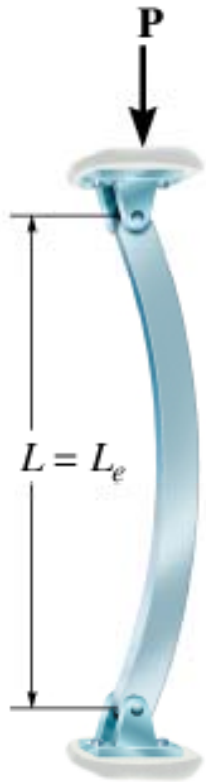
#### Effective length

- If a column is not supported by pinned-ends, then Euler's formula can also be used to determine the critical load.
- “ $L$ ” must then represent the distance between the zero-moment points.
- This distance is called the columns' effective length,  $L_e$ .

## 13. Buckling of Columns

### 13.3 COLUMNS HAVING VARIOUS TYPES OF SUPPORTS

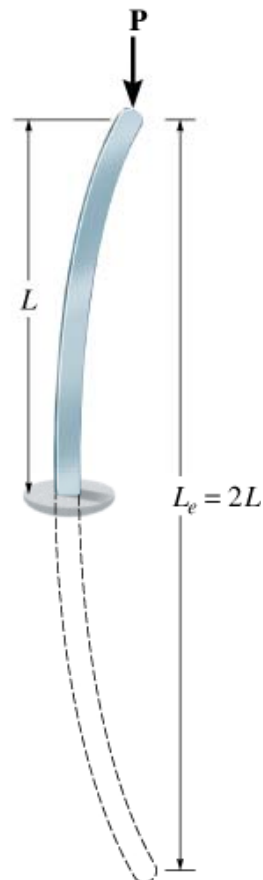
#### Effective length



Pinned ends

$$K = 1$$

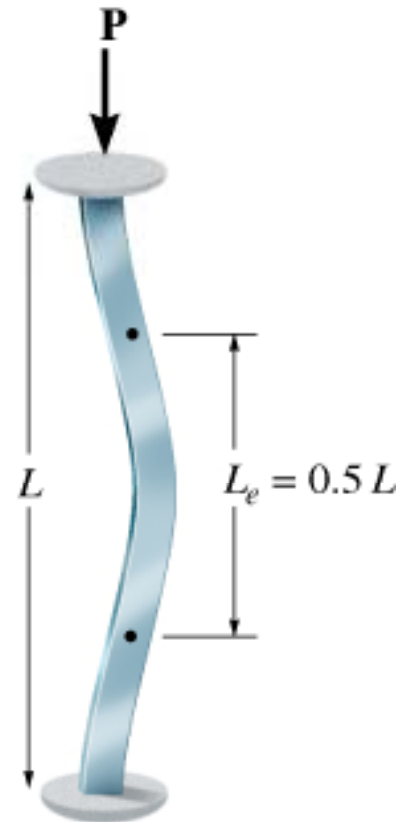
(a)



Fixed and free ends

$$K = 2$$

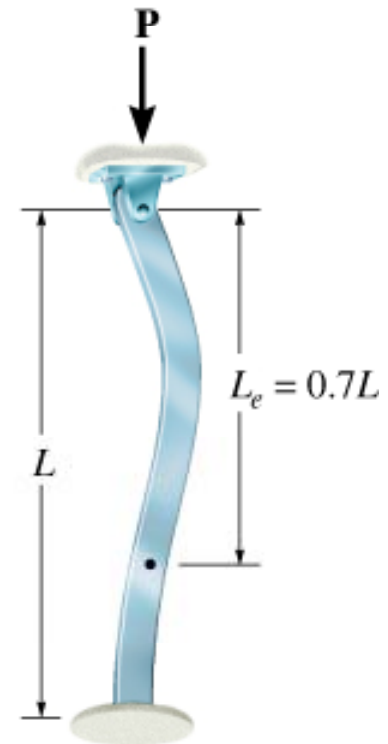
(b)



Fixed ends

$$K = 0.5$$

(c)



Pinned and fixed ends

$$K = 0.7$$

(d)

## 13. Buckling of Columns

### 13.3 COLUMNS HAVING VARIOUS TYPES OF SUPPORTS

#### Effective length

- Many design codes provide column formulae that use a dimensionless coefficient  $K$ , known as the effective-length factor.

$$L_e = KL \quad (13-10)$$

- Thus, Euler's formula can be expressed as

$$P_{cr} = \frac{\pi^2 EI}{(KL)^2} \quad (13-11)$$

$$\sigma_{cr} = \frac{\pi^2 E}{(KL/r)^2} \quad (13-12)$$

## 13. Buckling of Columns

### 13.3 COLUMNS HAVING VARIOUS TYPES OF SUPPORTS

#### Effective length

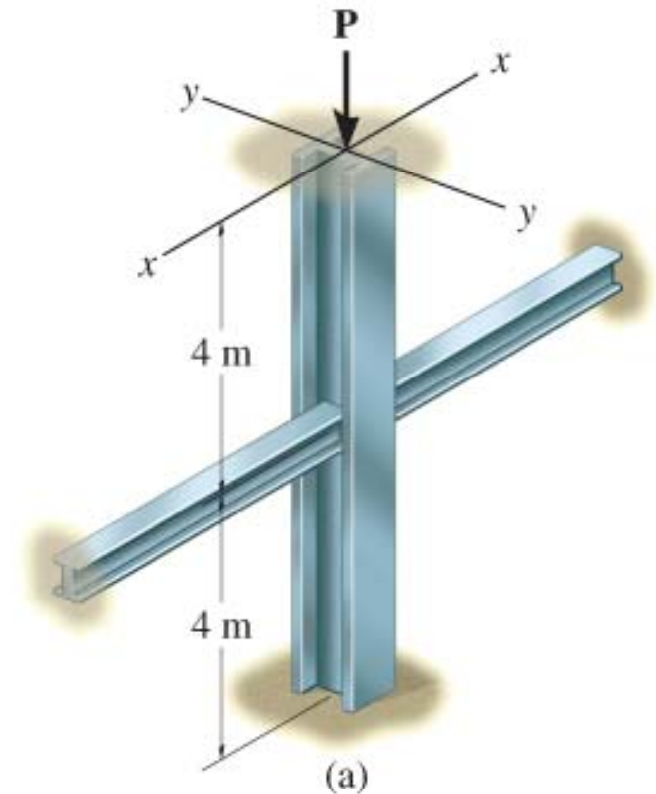
- Here  $(KL/r)$  is the column's effective-slenderness ratio.

## 13. Buckling of Columns

### EXAMPLE 13.2

A W150×24 steel column is 8 m long and is fixed at its ends as shown. Its load-carrying capacity is increased by bracing it about the  $y$ - $y$  axis using struts that are assumed to be pin-connected to its mid-height. Determine the load it can support so that the column does not buckle nor material exceed the yield stress.

Take  $E_{st} = 200 \text{ GPa}$  and  $\sigma_Y = 410 \text{ MPa}$ .





## 13. Buckling of Columns

### EXAMPLE 13.2 (SOLN)

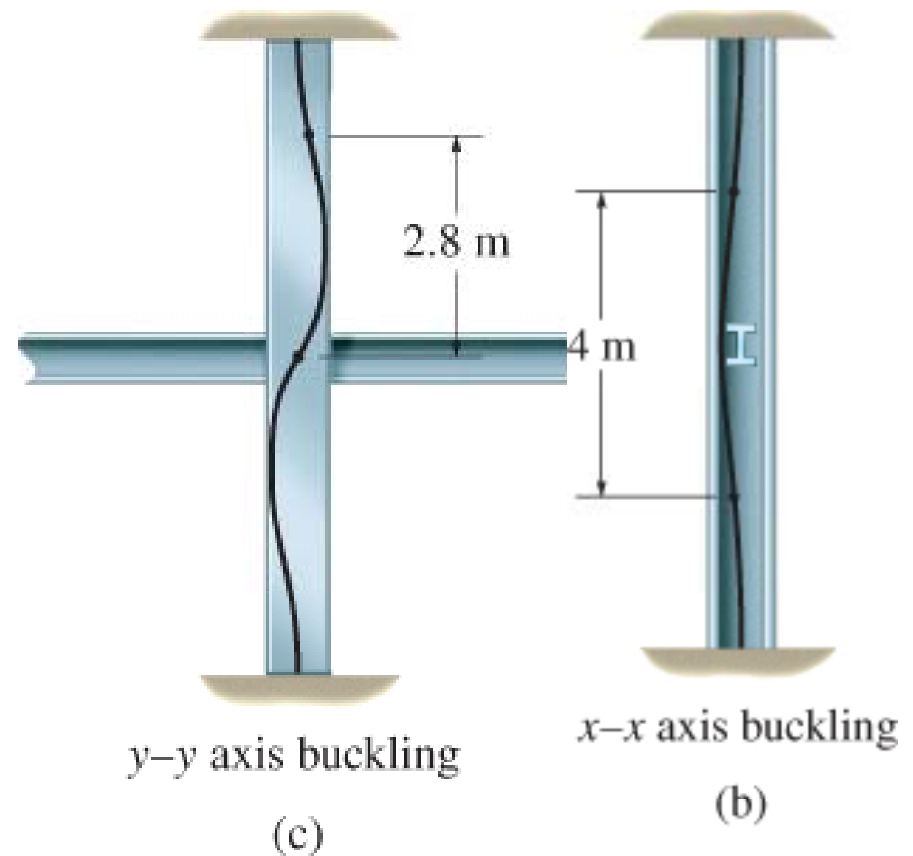
Buckling behavior is different about the  $x$  and  $y$  axes due to bracing.

Buckled shape for each case is shown.

The effective length for buckling about the  $x$ - $x$  axis is  $(KL)_x = 0.5(8 \text{ m}) = 4 \text{ m}$ .

For buckling about the  $y$ - $y$  axis,  $(KL)_y = 0.7(8 \text{ m}/2) = 2.8 \text{ m}$ .

We get  $I_x = 13.4 \times 10^6 \text{ mm}^4$  and  $I_y = 1.83 \times 10^6 \text{ mm}^4$  from Appendix B.



## 13. Buckling of Columns

### EXAMPLE 13.2 (SOLN)

Applying Eqn 13-11,

$$(P_{cr})_x = \frac{\pi^2 EI_x}{(KL)_x^2} = \frac{\pi^2 [200(10^6) \text{ kN/m}^2] 13.4(10^{-6}) \text{ m}^4}{(4 \text{ m})^2}$$

$$(P_{cr})_x = 1653.2 \text{ kN}$$

$$(P_{cr})_y = \frac{\pi^2 EI_y}{(KL)_y^2} = \frac{\pi^2 [200(10^6) \text{ kN/m}^2] 1.83(10^{-6}) \text{ m}^4}{(2.8 \text{ m})^2}$$

$$(P_{cr})_y = 460.8 \text{ kN}$$

By comparison, buckling will occur about the  $y$ - $y$  axis.

## 13. Buckling of Columns

### EXAMPLE 13.2 (SOLN)

Area of x-section is 3060 mm<sup>2</sup>, so average compressive stress in column will be

$$\sigma_{cr} = \frac{P_{cr}}{A} = \frac{460.8(10^3) \text{ N}}{3060 \text{ mm}^2} = 150.6 \text{ N/mm}^2$$

Since  $\sigma_{cr} < \sigma_Y = 410 \text{ MPa}$ , buckling will occur before the material yields.

## 13. Buckling of Columns

### EXAMPLE 13.2 (SOLN)

NOTE: From Eqn 13-11, we see that buckling always occur about the column axis having the largest slenderness ratio. Thus using data for the radius of gyration from table in Appendix B,

$$\left(\frac{KL}{r}\right)_x = \frac{4 \text{ m}(1000 \text{ mm/m})}{66.2 \text{ mm}} = 60.4$$

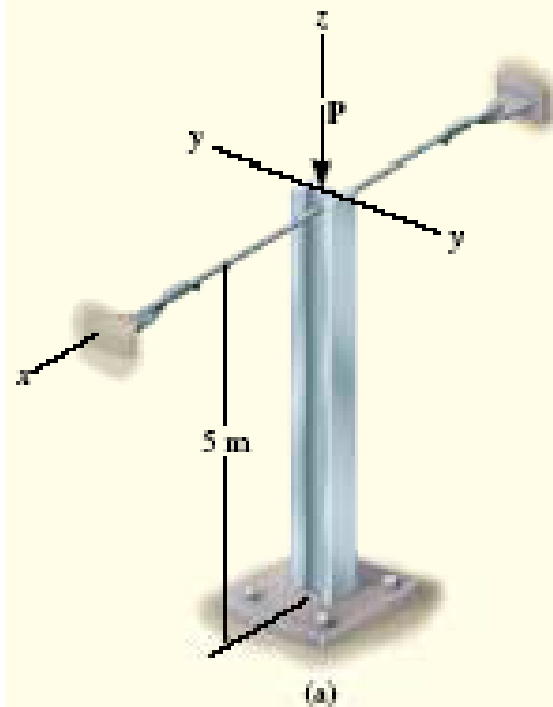
$$\left(\frac{KL}{r}\right)_y = \frac{2.8 \text{ m}(1000 \text{ mm/m})}{24.5 \text{ mm}} = 114.3$$

Hence,  $y$ - $y$  axis buckling will occur, which is the same conclusion reached by comparing Eqns 13-11 for both axes.

## 13. Buckling of Columns

### EXAMPLE 13.3 (SOLN)

The aluminum column is fixed at its bottom and is braced at its top by cables so as to prevent movement at the top along the  $x$  axis, Fig. 13–14a. If it is assumed to be fixed at its base, determine the largest allowable load  $P$  that can be applied. Use a factor of safety for buckling of  $F.S. = 3.0$ . Take  $E_a = 70 \text{ GPa}$ ,  $\sigma_Y = 215 \text{ MPa}$ ,  $A = 7.5(10^{-3}) \text{ m}^2$ ,  $I_x = 61.3(10^{-6}) \text{ m}^4$ ,  $I_y = 23.2(10^{-6}) \text{ m}^4$ .



### Solution

Buckling about the  $x$  and  $y$  axes is shown in Fig. 13-14*b* and 13-14*c*, respectively. Using Fig. 13-12*a*, for  $x$ - $x$  axis buckling,  $K = 2$ , so  $(KL)_x = 2(5 \text{ m}) = 10 \text{ m}$ . Also, for  $y$ - $y$  axis buckling,  $K = 0.7$ , so  $(KL)_y = 0.7(5 \text{ m}) = 3.5 \text{ m}$ .

Applying Eq. 13-11, the critical loads for each case are

$$\begin{aligned}(P_{\sigma})_x &= \frac{\pi^2 EI_x}{(KL)_x^2} = \frac{\pi^2 [70(10^9) \text{ N/m}^2] (61.3(10^{-6}) \text{ m}^4)}{(10 \text{ m})^2} \\ &= 424 \text{ kN} \\ (P_{\sigma})_y &= \frac{\pi^2 EI_y}{(KL)_y^2} = \frac{\pi^2 [70(10^9) \text{ N/m}^2] (23.2(10^{-6}) \text{ m}^4)}{(3.5 \text{ m})^2} \\ &= 1.31 \text{ MN}\end{aligned}$$

By comparison, as  $P$  is increased the column will buckle about the  $x$ - $x$  axis. The allowable load is therefore

$$P_{\text{allow}} = \frac{P_{\sigma}}{\text{F.S.}} = \frac{424 \text{ kN}}{3.0} = 141 \text{ kN} \quad \text{Ans.}$$

Since

$$\sigma_{\sigma} = \frac{P_{\sigma}}{A} = \frac{424 \text{ kN}}{7.5(10^{-3}) \text{ m}^2} = 56.5 \text{ MPa} < 215 \text{ MPa}$$

Euler's equation can be applied.

